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## Full length article On the impact toughness of gradient-structured metals

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### ABSTRACT

Gradient-structured (GS) materials are capable of displaying high strength without compromising ductility, which can result in damage-tolerant structures. However, due to the difficulties in fabricating bulk GS materials, there has been only limited studies on the fracture behavior in GS metals. In the present work, the impact toughness of the macroscale GS pure Ni plates was investigated using instrumented Charpy impact testing. The gradient orientation was found to have a significant influence on the impact toughness of GS Ni. For gradient structures that transition from coarse grains (CG) to nano-grains (NG), termed CG $\rightarrow$ NG gradients (in the present study from  $\sim 8 \,\mu$ m to  $\sim 30$  nm), the absorbed energy and the tensile strength were increased, respectively, by 1.6 and 2.3 times from those exhibited by uniform coarse-grained structures, demonstrating a simultaneous enhancement in strength and impact toughness. Analysis of load-displacement curves revealed that the resistance to both crack initiation and propagation were significantly enhanced as the crack penetrated through the CG->NG gradient structure, leading to markedly rising dynamic R-curve behavior estimated from nonlinearelastic fracture mechanics *J*-based measurements. The superior fracture resistance in the  $CG \rightarrow NG$  gradient structure was found to originate from sustained ductile fracture by microvoid coalescence, taking place not only in the initial CG zone, but also within the latter NG regions where adiabatic shear bands form during impact; in these latter regions, plasticity becomes enhanced due to grain coarsening induced by recrystallization under the dynamic loading. The present work not only reveals how the dynamic fracture resistance can be significantly enhanced in GS metals, but also provides structure-design strategies for developing superior metallic materials for impact engineering applications.

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### 1 1. Introduction

Achieving both high strength and high toughness is a longstand-2 ing objective of materials scientists in their efforts to improve the 3 mechanical performance of structural materials. Toughness, which 4 5 can be simply considered as an integration of strength and ductility [1,2], represents the material's resistance to fracture. Nevertheless, 6 strength and toughness tend to be mutually exclusive [1]. In many 7 cases, high-strength metals are characterized by low ductility and 8 inferior fracture toughness at room temperature [3–6]. For instance, 9 10 for an electrodeposited nanocrystalline Ni with an ultrahigh tensile strength of over 1.57 GPa, the reported fracture toughness ( $K_{Ic}$ ) is 35. 11 5 MPa  $m^{1/2}$  [6]; in contrast, the fracture toughness of the annealed 12 coarse-grained Ni is more than six times higher ( $K_{Ic} \sim 220$  MPa m<sup>1/2</sup>) 13 but with a far lower tensile strength of 340 MPa [4,7]. Similarly, the 14 15 impact energy of an ultrafine-grained Grade 5 Ti alloy (grain size  $\sim$ 

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300 nm) with a high tensile strength of 1435 MPa is  $0.15 \text{ J} \text{ mm}^{-2}$ , 16 which is significantly lower than the impact energy  $(0.54 \text{ J} \text{ mm}^{-2})$  of 17 a coarse-grained Grade 5 Ti alloy, but this latter material displays a 18 much lower tensile strength (965 MPa) [8]. This conflict between 19 strength and toughness leads to an inevitable compromise for 20 designing strong and tough structural materials [1,2].

Recently, gradient structures have attracted considerable atten- 22 tion. Akin to natural materials where graded structures are com-23 monly utilized [9], gradients have been creatively engineered into 24 novel structural materials to attain favorable combinations of 25 mechanical properties [10-14]. It has been shown, for example, that 26 high strength and good ductility can be simultaneously acquired in 27 such gradient structured (GS) materials [10-12,15-17], which sug- 28 gests a promising means of designing high strength, damage-tolerant 29 materials. We hypothesize that for a nominally ductile metal, a hard 30 phase in a GS material may act to inhibit crack initiation, whereas a 31 softer phase may enable optimized resistance to crack propagation. 32 Several numerical studies have been carried out to examine the influ-33 ence of gradient structures on the fracture resistance. As an example, 34 using finite element simulations, Cavaliere [18] found that, as a crack 35 propagates from the softer region to the harder region in a GS Ni-W 36

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2

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Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

alloy, the I-based fracture resistance increases. Moreover, molecular 37 dynamics simulations [19] predict that a large gradient in grain size 38 39 can serve as a barrier against crack propagation in nanocrystalline Ni, leading to a higher overall toughness. However, limited experimental 40 41 studies have focused on the toughness of *bulk* GS materials. The lack of experimental studies on the fracture behavior of GS metals can be 42 traced to difficulties in fabricating such gradient structures at macro-43 scale dimensions. Early studies on many GS materials [11,12,20,21] 44 45 involved section thicknesses of only a few hundred micrometers, which can rarely satisfy the size requirements for realistic fracture 46 47 toughness measurement (e.g., as per ASTM Standard 1820 [22]). Regardless of the technical difficulties in processing large-dimen-48 sion GS metals though, there are still several unanswered ques-49 tions about the toughness of such bulk materials, as to (i) 50 51 experimental confirmation that the fracture toughness can actu-52 ally be improved by introducing a gradient structure, and (ii) 53 whether there is a specific effect of a grain-size gradient in resist-54 ing crack growth under both static and dynamic loading. Clearly 55 these issues hinder a thorough understanding of toughening 56 mechanisms in GS metals and limit their potential applications in engineering service. While our recent work has revealed the 57 micro-mechanisms underlying the superior damage-tolerance 58 associated with crack initiation and propagation in bulk-sized 59 60 gradient nickel under quasi-static loading [23], the corresponding fracture behavior of gradient metallic materials subjected to 61 dynamic loading has not yet to be explored. Accordingly, the 62 objective of the present study is to investigate the influence of 63 the gradient structure on the initiation and propagation of cracks 64 65 in metals under the dynamic/impact loading conditions.

Specifically, this paper describes an experimental study to evalu-66 ate the impact toughness of gradient nickel using instrumented 67 68 Charpy impact testing. By analyzing the recorded load-displacement curves obtained under impact loading, the resistance to crack initia-69 70 tion and propagation is quantified from the absorbed energy and the 71 estimated dynamic crack-resistance R-curves. We systematically 72 investigate the effect of gradients in grain size on the impact tough-73 ness of GS Ni plates. We find that coarse to nano-grained (CG $\rightarrow$ NG) 74 gradients, with a reduction in grain size from  ${\sim}8~\mu{\rm m}$  to  ${\sim}30\,{\rm nm}$ along the crack path, displayed an impact toughness higher than 75 76 those of other gradient and monolithic-structured (MS) Ni structures.

Moreover, based on nonlinear-elastic *J*-integral measurements, we 77 observe a markedly rising dynamic R-curve for the  $CG \rightarrow NG$  gradient 78 structure, indicative of its superior damage-tolerance under impact 79 loading. Based on microstructural characterization, we further discuss the salient toughening mechanisms underlying such crack initiation and crack propagation behavior in the  $CG \rightarrow NG$  gradient Ni structure. 83

### 2. Experimental methods

### 2.1. Materials and test specimens

Bulk-sized gradient-structured (GS) Ni plates, with dimensions of 86  $\sim$ 40  $\times$  60  $\times$  2 mm<sup>3</sup> were fabricated by a direct-current (DC) electro-87 plating set-up. The plating bath composition and operating condi-88 tions are described elsewhere [10]. With an increase in the current 89 density from 10 to 100 mA/cm<sup>2</sup> and the additive (sodium saccharin) 90 concentration from 1 to 5 g/L, the grain size can be continuously 91 refined from several micrometers ( $\sim 8 \ \mu m$ ) to a few tens of nano-92 meters ( $\sim$ 30 nm) along the deposition direction. In order to fabricate 93 the bulk-sized nano- to coarse-grained (NG $\rightarrow$ CG gradient) and 94 coarse- to nano-grained (CG $\rightarrow$ NG gradient) specimens for Charpy 95 pendulum impact testing, a ~2 mm-thick layer of monolithic nano-96 grained (NG) Ni (grain size  $\sim$ 50 nm) was further coated on the NG 97 end and the CG end of the GS Ni plate, respectively, forming a GS 98 plate with the final dimensions of  $\sim 40 \times 60 \times 4 \text{ mm}^3$  (Figs. 1a, b). For 99 comparison purposes, monolithic electrodeposited nano-grained 100 (NG) and rolled CG Ni plates were also prepared (Fig. 1d). All the GS 101 and MS Ni plates were annealed at 393 K for 8 h to relieve any resid-102 ual stresses induced by electrodeposition before subsequent speci-103 men machining. 104

Miniature impact specimens with dimensions of  $3 \times 5 \times 55 \text{ mm}^3$ 105 were machined from the MS and GS Ni plates (Fig. 1c). A 45° V-notch 106 with a depth of  $\sim$ 1.5 mm was machined in the 3 mm-wide specimens. 107 The 1.5 mm-long ligament of the impact specimens, where the crack 108 initiates and propagates, is composed of a complete gradient or mono-109 lithic structure, *i.e.* involving a CG, CG $\rightarrow$ NG, NG, or NG $\rightarrow$ CG structure, 110 as shown in Fig. 1d. Before mechanical testing, the specimens were 111 processed with several rounds of mechanical and electropolishing, so 112 that smooth and residual-stress-free surfaces were prepared. 113



**Fig. 1.** Schematic diagrams illustrating the fabrication of the bulk-sized nano- to coarse-grained (NG $\rightarrow$ CG gradient) (a) and coarse- to nano-grained (CG $\rightarrow$ NG gradient) (b) impact specimens from the gradient-structured (GS) plate coated with nano-grained Ni layer. (c) The miniaturized 45 ° V-notched specimen for Charpy impact testing with the dimensions of 3 × 5 × 55 mm<sup>3</sup>. (d) Typical grain-size gradients ( $\sim 8 \mu m$  to  $\sim 30 nm$ ) with two gradient orientations (CG $\rightarrow$ NG gradient and NG $\rightarrow$ CG gradient) and two monolithic grain structures (CG and NG) in the ligament of the specimens as circumscribed by the dashed rectangular in (c). All dimensions are in millimeters.

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

179

#### 2.2. Mechanical characterization 114

#### 2.2.1. Microhardness/grain-size measurements 115

116 To determine the grain-size profiles along the gradient/monolithic 117 ligaments in the V-notched GS/MS specimens, microhardness profiles were measured across the GS/MS specimen ligaments using a Qness 118 Q10 A+ microhardness tester with Vickers indenter. An indentation 119 load of 5 g was applied with a dwell time of 10 s; separation between 120 any two neighboring indentations was  $\sim$ 50  $\mu$ m. Grain sizes were cal-121 culated from the microhardness values using the Hall-Petch relation-122 ship [24] as follows: 123

$$H = 2.17 \text{ GPa} + (16.11 \text{ GPa} \cdot nm^{1/2}) d^{-1/2}, \tag{1}$$

124 125

where *H* is the microhardness value and *d* represents the grain size.  
The validation of determining the grain-size distribution from the  
microhardness profile by using Eq. 
$$(1)$$
 was checked with direct meas-

127 urements by x-ray diffraction (XRD) and SEM imaging (for further 128 129 information, see Fig. A.1 in the Supplementary Data in ref. [23]).

#### 2.2.2. Quasi-static uniaxial tensile testing 130

To determine the quasi-static yield strength and tensile strength 131 of the GS and MS structures, the same structured plates with a thick-132 ness of  $\sim$ 0.5 mm were prepared. The plates were then cut into dog-133 bone shaped tensile specimens, with a total length of 25 mm, a gage 134 cross-section of  $0.5 \times 1.2 \text{ mm}^2$  and a gage length of 6 mm. Uniaxial 135 136 tensile tests were performed on an Instron-5848 micro-tester system (Instron Corporation, Norwood, MA, USA) in ambient air, at a strain 137 rate of  $3.0 \times 10^{-4}$  s<sup>-1</sup>. Three valid tensile tests were repeated for 138 each structured condition. Before tensile testing, all tensile specimens 139 were electropolished to acquire a smooth, stress-free surface. 140

#### 2.2.3. Instrumented Charpy impact testing 141

The instrumented Charpy pendulum impact testing was per-142 143 formed on a Zwick RKP 450 pendulum impact tester (ZwickRoell, Germany) in ambient air. The impact striker and the experimental 144 145 procedure of the impact experiments were in accordance with ASTM Standard E23-18 [25]. The loading span was 40 mm. The impact 146 speed was 5.23 m s<sup>-1</sup>. At least three valid Charpy impact tests were 147 performed on each structured condition to record the dynamic load-148 149 displacement curves.

The impact toughness of the various MS and GS structures was 150 evaluated in terms of the total absorbed impact energy,  $W_{\text{total}}$ , *i.e.*, the 151 total area under the load-displacement curve recorded during the 152 impact testing (Fig. 2a). We further analyzed the absorbed impact 153 energies associated with different stages of crack growth, *i.e.*, crack 154 initiation, stable crack growth, unstable crack growth, and final col-155 lapse, following the methods described in refs. [27-29]. The four fea-156 tured cracking stages correspond to four sections on a typical impact 157 load-displacement curve, as schematically illustrated in Fig. 2a. The 158 loads P<sub>gy</sub>, P<sub>in</sub>, P<sub>m</sub>, P<sub>iu</sub> and P<sub>a</sub> in Fig. 2a correspond to the general yield 159 load, the load at crack initiation, the maximum load, the load at the 160 start of unstable crack growth, and the load at the end of unstable 161 crack growth, respectively. The load  $P_{gy}$  corresponds to the onset of 162 the inelastic segment of the load-displacement curve, indicating the 163 yielding of the impact specimen. The crack initiation point, P<sub>in</sub> was 164 determined by applying the "compliance changing rate" (CCR) 165 method [26] (for details, see Appendix A). The compliance changing 166 rate ( $\Delta C/C$ ) is defined as follows: 167

$$\Delta C/C = (C - C_{el})/C_{el}, \tag{2}$$

where  $C = d\Delta/dP$  is the secant compliance ( $\Delta$  and P, respectively, 169 denote the displacement and the load), and  $C_{el}$  is the elastic compli-170 ance, which is determined as the slope  $(P_{\rm el}/\Delta_{\rm el})$  of the linear elastic 171 section of the load-displacement curve (Fig. 2b). The crack initiation 172 corresponds to the transition point at which an abrupt change in the 173 value of  $\Delta C/C$  is detected from the curve of  $\Delta C/C$  vs. displacement 174 (Fig. 2b). The section of unstable crack growth refers to the quasi-lin-175 ear section of the load-displacement curve between  $P_{iu}$  and  $P_{a}$ , where 176 a steep drop of the load is shown (Fig. 2a); the remaining section after 177 point  $P_a$  corresponds to the final collapse stage. 178

#### 2.3. Microstructure characterization

Prior to mechanical testing, the microstructures along the liga-180 ments in the GS and MS impact specimens were characterized using 181 an FEI Nova NanoSEM 430 scanning electronic microscope (SEM) 182 operated at a voltage of 15 kV using a Low-kV, High-Contrast detector 183 (vCD) in the backscattered-electron (BSE) imaging mode. 184

To specifically examine the crack-path profile and discern the 185 deformation and failure mechanisms in the vicinity of the crack tip 186 and crack wake, we electrodeposited a  $\sim$ 0.5 mm thick Ni layer on the 187 exterior of the impact-tested specimens, so that the fracture surface 188



Fig. 2. (a) Schematic diagram showing the absorbed energies associated with different stages of fracture (crack initiation, stable crack growth, unstable crack growth, collapse) during an instrumented impact test. Here, Pgy, Pin, Pm, Piu and Pa correspond, respectively, to the general yield load, the load at crack initiation, the maximum load, the load at the start of unstable crack growth, and the load at the end of unstable cracking. (b) Demonstration of the determination of the load at crack initiation P<sub>in</sub> using the compliance changing rate (CCR) method [26]. Here, the load-displacement curve is obtained for the NG $\rightarrow$ CG gradient structure.

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4

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

was protected. The impact-tested specimens were then sliced through 189 the thickness at the mid-section in order to examine the deformation 190 and fracture processes under plane-strain conditions. The interior sur-191 face of one sliced half was progressively polished to a mirror finish, fol-192 lowed by a final round of electropolishing. The microstructure along 193 the crack path on the mid-plane section was imaged using FEI Nova 194 NanoSEM 460 SEM operated at a voltage of 15 kV with a Circular Back-195 Scattered (CBS) detector in the BSE imaging mode. Additional micro-196 structure characterizations using electron backscatter diffraction 197 (EBSD) analysis were performed for the CG and NG $\rightarrow$ CG specimens to 198 199 examine the impact-deformed grain structures in the plastic-wake region and in the uncracked ligament, respectively. EBSD scans for the 200 CG specimens were conducted using an FEI Nova NanoSEM 460 SEM 201 equipped with an Oxford EBSD detector, operating at 20 kV with a 202 203 step size of 200 nm. For the NG→CG specimens, EBSD scans were conducted on a Zeiss Supra 55 microscope attached with an Oxford EBSD 204 205 detector, operating at 15 kV with a step size of 50 nm.

### 207 **3. Results**

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### 208 3.1. Microstructure of the impact specimens

BSE images taken from pure NG and pure CG structures confirmed a uniform nano-sized grain structure (Fig. 3a) and a uniform equiaxed coarse-grained structure (Fig. 3b), respectively. Based the direct measurement from the BSE images taken from the CG specimens, the 212 average grain size of the pure CG structure was estimated to be 213  $\sim$ 50  $\mu$ m. BSE images taken from the 1.5 mm-long ligaments revealed 214 that the grain-size gradient transitions smoothly from nano-grains to 215 coarse grains and *vice versa*, respectively, for the NG $\rightarrow$ CG and 216 CG $\rightarrow$ NG specimens (Figs. 3c, d). 217

To characterize the grain size of the pure NG structure and the 218 grain-size profiles of the GS structures, microhardness profiles were 219 measured along the crack-propagation direction as a function of the 220 normalized distance from 0 to 1 (0: notch tip; 1: back-end of the liga-221 ment) (Fig. 3e). The microhardness of the pure NG specimens were 222 found to vary slightly, increasing by  $\sim 10\%$  from  $\sim 4.5$  to  $\sim 5.0$  GPa 223 along the crack-propagation direction. This microhardness variance 224 is likely due to the slight variance of grain size caused by the degrada-225 tion of the additives during the long-time electroplating. Using the 226 Hall-Petch relationship (Eq. (1)), the average grain size of the pure 227 NG specimens was estimated to be  $\sim$ 40 nm, although strictly speak-228 ing, the grain size varied from 48 nm to 32 nm from the notch tip to 229 the back-face of the ligament. 230

For the NG $\rightarrow$ CG gradient specimens, the microhardness continuously decreased from  $\sim$ 5.1 to  $\sim$ 2.4 GPa as the normalized distance 232 shifts from 0 (at the notch tip) to 1 (at the back-face of the specimen). 233 In contrast, the microhardness for the CG $\rightarrow$ NG gradient specimens 234



**Fig. 3.** Typical backscattered-electron (BSE) images taken from the  $\sim$ 1.5 mm-long ligament in different specimens revealing the various grain structures: (a) pure NG, (b) pure CG, (c) gradient NG $\rightarrow$ CG, and (d) gradient CG $\rightarrow$ NG, respectively. (e) Microhardness profiles measured on the ligaments of various structured impact specimens as a function of the normalized distance ranging from 0 to 1 (on the top of the figure), indicating the location ranging from the notch tip to the back-face of the specimen. (f) Grain-size distribution profiles calculated from the microhardness profiles using the Hall–Petch relationship shown in Eq. (1). Note that the grain size of the pure CG sample is measured from the BSE micrographs.

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increased from  $\sim$ 2.2 to  $\sim$ 5.0 GPa with the increasing normalized dis-235 tance. The hardness profiles of the NG $\rightarrow$ CG and CG $\rightarrow$ NG specimens 236 237 are roughly symmetrical, reflecting that the profiles of their grain-size distributions are approximately identical. Using the 238 239 Hall–Petch relationship (Eq. (1)), the grain size profiles for the GS specimens were calculated as a function of the normalized distance 240 as shown in Fig. 3f. A smooth transition of the grain size from 241  $\sim$ 30 nm to  $\sim$ 8  $\mu$ m without sharp interfaces is characterized in the 242 NG $\rightarrow$ CG specimen whereas the CG $\rightarrow$ NG specimen exhibited approx-243 imately identical grain-size distribution but in opposite direction. 244

### 245 3.2. Quasi-static tensile properties

246 The representative engineering stress-strain curves of the gradient  $(CG \rightarrow NG \text{ or } NG \rightarrow CG)$  and monolithic (NG and CG) structures were mea-247 sured using room-temperature uniaxial tensile tests at a strain rate of 248  $3 \times 10^{-4} s^{-1}$ ; results are presented in Fig. 4 and summarized in Table 1. 249 Results show that the tensile strength ( $\sigma_{\mathrm{uts}}$ ) of the GS structure at 250  $\sim$ 760 MPa is far higher, by 126%, than that of the uniform coarse-grained 251 CG structure (where  $\sigma_{\rm uts}$   $\sim$  336 MPa), but much lower, by 43%, than that 252 of the uniform nano-grained NG structure (where  $\sigma_{
m uts} \sim 1340$  MPa). In 253 254 contrast to the strength, the total elongation ( $\varepsilon_f$ ) of the GS structure at  $\sim$ 12% is 67% smaller than that of the CG material (where  $\varepsilon_f \sim$  37%), but 255  $\sim$ 100% larger than that of the NG material (where  $\varepsilon_f \sim 6\%$ ). 256

Results from the tensile tests show that there is a trade-off between 257 258 strength and ductility in the CG and NG structures; the high-strength 259 NG material displays brittle-like behavior with one-sixth of the ductility, whereas the ductile CG material has a factor of four lower tensile 260 strength. However, consistent with other gradient- and heteroge-261 neous- structured materials [10-12,30-33], the combination of 262 strength and ductility in gradient GS materials is far superior to that in 263 264 uniform-grained materials, regardless of their specific grain size. This 265 strongly implies that the use of gradient architectures can reliably give 266 rise to a favorable synergy of strength and damage-tolerance.



The load-displacement response of the impact specimens 268 obtained from the instrumented Charpy impact tests are presented 269 in Fig. 5. It can be seen that the impact event in the uniform NG speci-270 mens results in an initial increase in load, which then decreases rap-271 idly as fracture ensues; a maximum load of 567 N is reached at the 272 displacement of  $\sim$ 2.5 mm. This indicates that the NG structure has a 273 strong resistance to crack initiation, but once a crack is formed, it 274 propagates catastrophically with little resistance to crack growth. 275 Conversely, the uniform CG specimens display a slower rising load, 276 but only up to a maximum load of 185 N, *i.e.*, a factor of three smaller 277 than the NG material, but at a four-fold increase in displacement of 278  $\sim$ 10 mm. Beyond the maximum load, the load drops gradually, indi-279 cating a opposite response to the NG material. Specifically, the CG Ni 280 structure displays a much lower resistance to crack initiation than 281 the NG Ni, but a much-improved resistance to crack growth. 282

Unlike the monolithic structures, it is evident from the load-dis-283 placement curves that the gradient orientation has a marked effect 284 on the impact resistances of the metallic Ni structures. For the 285 NG $\rightarrow$ CG specimens, the maximum load (where  $P_m \sim 358$  N) was 286 much lower than the 567 N for the NG specimens, presumably due to 287 the presence of coarse grains in the specimen ligament. The subse-288 quent rapid load drop results from rapid crack propagation which is 289 detrimental to the toughness. With respect to the  $CG \rightarrow NG$  speci-290 mens, the maximum load (where  $P_m \sim 362 \text{ N}$ ) was much higher than 291 the 185 N displayed by the CG Ni specimens. Similar to the CG speci-292 mens, the post-peak load drop was relatively slow in the  $CG \rightarrow NG$ 293 specimens, suggesting more resistance to crack propagation, which is 294 beneficial to the impact toughness. 295

### 3.4. Absorbed impact energy



**Fig. 4.** Representative engineering stress-strain curves of the monolithic (CG and NG) and gradient (NG $\rightarrow$ CG or CG $\rightarrow$ NG) Ni structures, obtained at room temperature under uniaxial tension at a strain rate of 3 × 10<sup>-4</sup> s<sup>-1</sup>.

The total absorbed impact energy (including both elastic and plastic 297 contributions) for the impact tests were determined by measuring the 298



**Fig. 5.** Load-displacement curves for the NG,  $NG \rightarrow CG$ ,  $CG \rightarrow NG$ , and CG V-notched specimens obtained from instrumented Charpy impact tests at room temperature. Insets show the corresponding impact-tested specimens, respectively.

Table 1

Quasi-static tensile properties of the GS and MS Ni structures at room temperature.

Specimen	Yield strength, $\sigma_{\rm y}({\rm MPa})$	Tensile strength, $\sigma_{\rm uts}$ (MPa)	Uniform elongation, $\epsilon_{ue}$	Elongation to failure, $\epsilon_{\rm f}$
NG NG $\rightarrow$ CG or CG $\rightarrow$ NG CG	$\begin{array}{c} 936\pm 53\\ 467\pm 14\\ 119\pm 5\end{array}$	$\begin{array}{c} 1341 \pm 33 \\ 758 \pm 14 \\ 336 \pm 2 \end{array}$	$\begin{array}{c} 3.97 \pm 0.32\% \\ 6.49 \pm 0.16\% \\ 28.05 \pm 1.95\% \end{array}$	$\begin{array}{c} 6.03 \pm 0.62\% \\ 12.22 \pm 0.51\% \\ 36.58 \pm 0.98\% \end{array}$

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**Fig. 6.** (a) Total absorbed impact energy (*W*), and (b) absorbed impact energies associated with different stages of crack initiation  $W_{in}$ , stable growth  $W_{stable}$ , and unstable growth  $W_{unstable}$  for the NG, NG $\rightarrow$  CG, CG $\rightarrow$ NG, and CG specimens, subjected to instrumented Charpy impact testing.

Table 2

The total absorbed impact energy and absorbed impact energies of various impact specimens corresponding to the different stages of fracture under dynamic loading.

Structures	W(J)	$W_{in}(J)$	W <sub>stable</sub> (J)	W <sub>unstable</sub> (J)
$\begin{array}{l} NG \\ NG \rightarrow CG \\ CG \rightarrow NG \\ CG \end{array}$	$\begin{array}{c} 1.76 \pm 0.17 \\ 1.34 \pm 0.34 \\ 5.07 \pm 0.45 \\ 3.14 \pm 0.20 \end{array}$	$\begin{array}{c} 0.45 \pm 0.07 \\ 0.32 \pm 0.03 \\ 0.36 \pm 0.03 \\ 0.14 \pm 0.06 \end{array}$	$\begin{array}{c} 0.65 \pm 0.04 \\ 0.30 \pm 0.04 \\ 2.25 \pm 0.09 \\ 2.16 \pm 0.08 \end{array}$	$\begin{array}{c} 0.34 \pm 0.06 \\ 0.29 \pm 0.10 \\ 1.77 \pm 0.04 \\ 0.73 \pm 0.09 \end{array}$

total area under the load-displacement curves; these data are summa-299 rized in Fig. 6a and Table 2. The CG specimens had an impact energy of 300 301  $3.14 \pm 0.20$  J, almost twice as that in the NG specimens,  $1.76 \pm 0.17$  J. The impact energy of the NG $\rightarrow$ CG specimens (1.34 $\pm$ 0.34]) was 302 303 approximately at the same level as that of the NG specimens, suggesting that the impact toughness of the latter structures is not necessarily 304 improved by creating a  $NG \rightarrow CG$  gradient structure. Conversely, the 305  $CG \rightarrow NG$  specimens exhibited a much higher total absorbed energy 306 than the other structures. Specifically, the total absorbed energy for the 307 308  $CG \rightarrow NG$  specimens  $(5.07 \pm 0.45 \text{ J})$  was 1.6, 2.9 and 3.8 times higher than that measured in the CG, NG and NG $\rightarrow$ CG specimens, respectively. 309 One of the advantages of using instrumented Charpy impact test-310 ing is that the energies consumed during different stages of fracture, 311 *i.e.*, crack initiation and propagation, can be separately estimated by 312 analyzing the recorded load-displacement curves. In order to quan-313 tify the crack resistance during the different cracking stages, the 314 absorbed energies,  $W_{in}$ ,  $W_{stable}$ , and  $W_{unstable}$ , associated, respectively, 315 with the stages of crack initiation, stable crack growth, and unstable 316 crack growth, were estimated, using the CCR method [26], by finding 317 the areas under the corresponding sections of the load-displacement 318 curve, as schematically illustrated in Fig. 2a. Results are presented in 319 320 Fig. 6b and listed in Table 2.

321 As shown in Fig. 6b, the NG specimens had a higher crack-initiation 322 energy ( $W_{in} = 0.45 \pm 0.07$  J) but much reduced energies for stable  $(W_{\text{stable}} = 0.65 \pm 0.04 \text{ J})$  and unstable  $(W_{\text{unstable}} = 0.34 \pm 0.06 \text{ J})$  crack 323 growth, compared to those measured for the CG specimens, where 324  $W_{in} = 0.14 \pm 0.06$  J,  $W_{stable} = 2.16 \pm 0.08$  J and  $W_{unstable} = 0.73 \pm 0.09$  J. 325 For the NG $\rightarrow$ CG specimens, where the crack initiated on the NG side 326 327 and propagated through the NG $\rightarrow$ CG gradient, the values of  $W_{in}$  (0.32  $\pm$ 0.03 J) and  $W_{\text{stable}}$  (0.30 ± 0.04 J) were only ~70% and ~46% of those in 328 the NG specimens, respectively. As such, compared to the uniform NG 329 330 material, the resistance to either crack initiation or crack propagation is not enhanced by creating the NG $\rightarrow$ CG gradient structure. Conversely, 331 332 for the  $CG \rightarrow NG$  specimens, where crack initiated on the CG side and propagated through the  $CG \rightarrow NG$  gradient, the crack initiation energy 333  $(W_{\rm in}$  = 0.36  $\pm$  0.03 J) was increased by ~160% compared to that in the 334 CG specimens (Fig. 6b). This result implies that the crack initiation 335

resistance of the CG structure can be effectively enhanced by creating a 336 coarse- to nano-grained gradient. Compared to the CG specimen, the 337 crack propagation resistance in the CG $\rightarrow$ NG structure was also 338 improved, which is evident by the increased values, by  $\sim$ 4.4% and 339  $\sim$ 143%, respectively, of the energies for stable and unstable crack propagation, compared to those for the uniform-grained CG specimens. 341

#### 3.5. *Plane-strain crack-path profiles*

In order to understand the difference in fracture resistance of the 343 various monolithic and gradient structures, it is necessary to discern the 344 fracture mode as well as the associated deformation mechanisms under 345 plane-strain conditions during crack propagation under dynamic load-346 ing. For this purpose, we sliced the impact-tested specimens through 347 the thickness at the mid-plane section. The crack-path profiles and the 348 deformed microstructures in the plastic-wake regions close to the crack 349 path were examined by SEM imaging in the BSE mode. Additional 350 microstructural characterization was performed for the gradient speci-351 mens in regions ahead of the crack tip within the uncracked ligament. 352

#### 3.5.1. Nano-grained NG structures

Fig. 7 presents the crack-path profile on the mid-plane section in the 354 fractured half for the pure NG structure. Shown is the fracture that fol-355 lowed a nominal mode-I crack path from the notch tip to the specimen 356 back-face. Comparing the BSE images of the microstructure taken at the 357 regions close to the crack path (regions C1-C3 in Fig. 7) to those taken 358 at remote regions that are more than 1 mm away (regions R1-R3 in 359 Fig. 7), no noticeable difference in the morphology and size of the nano-360 grains was detected. This indicates that the pure NG structures experi-361 enced a fine-scale brittle-like fracture during the dynamic loading. 362

### 3.5.2. Coarse-grained CG structures

In contrast to the pure NG structure, the crack-path profile at the 364 mid-section in the statistically torn half of a pure CG specimen 365 appeared to be more curved (Fig. 8a). Also visible are micro-voids 366 located at the plastic-wake region close to the crack path (see the 367 blue solid arrows in Fig. 8a). This is of typical indication that the 368 dynamic crack propagation in the pure CG structure was driven by 369 ductile fracture accommodated by microvoid coalescence. High-mag-370 nification BSE images (Figs. 8b,d) taken at regions C1, C3 in Fig. 8a 371 close to the dynamic crack path clearly show the formation of distinct 372 dislocation slip bands and sub-grain structures consisting of micro-373 and nano-sized dislocation cells. EBSD crystal orientation map 374 (Fig. 8c) scanned at region C2 adjacent to region C1 in Fig. 8a near the 375 crack path reveals that sub-grain dislocation cells form low-angle 376 boundaries with mis-orientation angles smaller than 15° (delineated 377 by tiny gray lines distributed discontinuously within the coarse 378 grains in Fig. 8c). By comparison, the original coarse-grain boundaries 379

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx



**Fig. 7.** Crack-path profile captured on the mid-plane section in the fractured half of a pure NG specimen. Fine-scale brittle-like fracture was seen in the pure NG structure under impact loading as evidenced by the fact that no apparent difference in the grain morphology and size was detected between the nano-grains at the regions close to the crack path (regions C1-C3) and those at remote regions more than 1 mm away (regions R1-R3). The dynamic crack-path profile is delineated in white dotted line. The propagation direction of the dynamic crack path is indicated by the red dotted arrows.



**Fig. 8.** Crack-path profile captured on the mid-plane section in the statically-torn half of the pure CG specimen. (a) Overall curved crack-path profile with microvoids in existence nearby (indicated by the blue solid arrows). This indicates ductile fracture by microvoid coalescence acted as the fracture mode in the pure CG structure. (b) Magnified regions C1 in (a) show the development of sub-grain structures – dislocation slip bands and dislocation cells resulting from dynamic plastic deformation. (c) EBSD crystal orientation map scanned at region C2 adjacent to region C1 in (a) near the crack path reveals sub-grain biolocation cells form low-angle boundaries (the tiny gray lines distributed discontinuously within the grain) with misorientation smaller than 15° The original coarse grain boundaries with a misorientation larger than 15° are delineated by black solid lines. (d) Magnified regions C3 in (a) indicates that the size of sub-grain dislocation cells was further refined as the crack extends to the latter half of the ligament. The dynamic crack-path profile is delineated in white dotted line in (a) and in red solid line in (b,d). The propagation direction of the dynamic crack path is denoted by the red dotted arrows.

were characterized to have misorientation angles larger than 15° 380 (delineated by black solid lines in Fig. 8c). The refined dislocation 381 substructures including dislocation slip bands and dislocation cells 382 are typically a result of dynamic plastic deformation (DPD) under 383 high strain-rate loading, consistent with the grain structures devel-384 oped near the blunting edge of coarse-grained copper after Charpy 385 386 impact test [34]. Moreover, along the crack-propagation direction, the average size of the sub-grain dislocation cells was reduced from 387

 ${\sim}1~\mu{\rm m}$  at locations about one quarter of the crack length (region C1 388 in Fig. 8a, as magnified in Fig. 8b), to  ${\sim}600~\mu{\rm m}$  at locations of about 389 three quarters of the crack length (region C3 in Fig. 8a, as magnified 390 in Fig. 8d), revealing an intensified effect of sub-grain refinement 391 associated with the dynamic crack extension. By examining whether 392 the original coarse grains contain the sub-grained slip bands and/or 393 dislocation cells, we can roughly estimate the depth of dynamic plassic-deformation zone (*i.e.* the distance extended from the crack path) 395

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7

8

### **ARTICLE IN PRESS**

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx



**Fig. 9.** Crack-path profile captured on the mid-plane section in the NG $\rightarrow$ CG gradient specimens. The brittle-like cracking mode that governed the fracture process within the initial pure nano-grain zone continues to dominate throughout the originally coarser-grained end of the NG $\rightarrow$ CG gradient zone due to the recrystallized grain refinement occurring in the adiabatic shear bands (ASBs), which are formed in the originally coarser-grained end of the NG $\rightarrow$ CG gradient zone. Regions C1-C3 reveal the microstructure resulted from dynamic plastic deformation in the plastic-wake regions along the crack path. Region C4 shows the refined lamellar structures developed in ASBs formed near the ligament end. By comparison, the representative original grain structures corresponding to the deformed microstructures in regions C1-C4 are captured in remote regions R1-R4 far away from the crack path. The dynamic crack-path profile is delineated in white dotted line. The propagation direction of the dynamic crack path is denoted by the red dotted arrows. The ASB region formed in the uncracked ligament is circumscribed by blue dotted curves. The striking end of the Charpy impact specimen is delineated by green dotted line.

being varied from approximately  $150 \,\mu$ m (near the notch tip) to approximately  $550 \,\mu$ m (near the crack end); this large region of local plasticity enables effective dissipation of the impact energy, which enhances resistance to crack advance, consistent with the high levels of the absorbed energy monitored during the various stages of crack propagation in the pure CG structure (Fig. 6a).

### 402 3.5.3. Gradient NG $\rightarrow$ CG structures

For the gradient structures, impact-tested NG $\rightarrow$ CG gradient speci-403 mens were significantly deformed, forming a V-shaped profile with a 404 wide crack opening angle of  $\sim 90^{\circ}$  (see the insect in the top right corner 405 406 of Fig. 9). The entire crack path was seen to span from the notch tip, through the pure nano-grain zone and the NG $\rightarrow$ CG gradient zone, 407 before terminating at a location roughly 70% of the total ligament 408 409 length. Following crack initiation, the initial  $\sim$ 250  $\mu$ m crack path within the nano-grain zone appeared to be straight and smooth. Com-410 paring the deformed microstructure in region C1 in Fig. 9 to the original 411 microstructure imaged at a region R1 remote ( $\sim$ 120  $\mu$ m away) from 412 the fracture surface, no obvious changes in grain morphology and size 413 414 could be seen, although a limited accumulation of shear bands was 415 observed in the nano-grains close to the crack path. These features indicate that a brittle, almost cleavage-like, fracture<sup>1</sup> with very restricted 416

plastic deformation dominates the initiation and early growth of the 417 crack within the nano-grain zone. This brittle fracture mode was mani-418 fested in the low levels of the crack initiation energy for the NG $\rightarrow$ CG 419 structure (Fig. 6b and Table 2). Once the crack entered the NG $\rightarrow$ CG gra-420 dient zone, *i.e.*, beyond region C1 (at the left half of Fig. 9), although the 421 grain size gradually increased, the crack profile maintained a brittle-422 like smooth and straight appearance until the grain size exceeded 423  $\sim$ 70 nm. With subsequent extension into the NG $\rightarrow$ CG gradient zone, 424 the brittle-like straight crack path transitioned into to a more tortuous 425 trajectory, although a nominal mode-I fracture was still maintained. 426

Of note here was that microstructural changes induced by dynamic 427 plastic deformation could clearly be identified in the plastic wake close 428 to the crack path. Specifically, in region C2 (at the mid of Fig. 9) of the 429 plastic-wake, where the original grain size was  $\sim$ 100 nm, traces of dis-430 location slip bands were evident; these were not in the original micro-431 structure, as imaged in a region such as R2 remote ( $\sim$ 220  $\mu$ m away) 432 from the crack surface. However, this dynamic plastic-deformation 433 zone is very narrow in size, extending to a depth of only a few tens of 434 micrometers below the crack surface, implying that energy-dissipation 435 due to plastic deformation and its contribution to fracture resistance 436 are still limited, even for grain sizes exceeding  $\sim$ 70 nm. 437

The dynamic crack terminated in region C3 (Fig. 9), leaving  $\sim$ 30% 438 of the original ligament uncracked. A high-magnification BSE image 439 taken from the crack tip showed abundant formation of equiaxed 440 ultra-fine grains with average sizes in the range of  $\sim$ 100 to  $\sim$ 500 nm, 441 although the original grain structure at this location had a columnar 442 shape with a width of  $\sim 1 \,\mu m$  (see region R3 about 300  $\mu m$  away 443 from the crack tip). The transformation of the 1  $\mu$ m-wide columnar 444 grains to equiaxed ultra-fine grains is evidence of the occurrence of 445 dynamic recrystallization of the grain structures in the vicinity of the 446

<sup>&</sup>lt;sup>1</sup> As nickel has a face-centered cubic structure, it would not normally experience a ductile-to-brittle transition with a brittle fracture mechanism such as cleavage fracture, as is common in body-centered cubic materials, such as ferritic iron. Nickel tends to fail by microvoid coalescence at both low and high temperatures. Accordingly, the terms "brittle" and "brittle-like" are used here, not with reference to a fracture mechanism *per se*, but rather to a low-energy fracture showing little evidence, at the scale of observation, of incumbent plastic deformation, and which is characterized by a low fracture toughness.

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

492

crack during impact loading [35–37]. We presume that such recrystallization originated from the abrupt adiabatic increase in temperature during the impact loading as there would be insufficient time to
dissipate heat converted from the work of plastic deformation during
the extremely short impact loading times [34,37].

In addition to the dynamic recrystallization in the vicinity of the 452 crack tip, adiabatic shear bands (ASBs) were visible in a 125  $\mu$ m wide 453 region in the uncracked ligament spanning all the way from the crack 454 tip to the striking end, *i.e.*, the region bounded by blue dotted curves in 455 Fig. 9. High-magnification BSE imaging, with an EBSD crystal orienta-456 457 tion map, of this region (C4 in Fig. 9) clearly reveals that the shear bands were composed of elongated lamella structures with an average 458 thickness of  $\sim$ 300 nm. Compared to the original coarse grains in the 459 remote region R4 outside the ASBs ( $\sim$ 200  $\mu$ m away), the average width 460 461 of the columnar-shaped grains was diminished under dynamic loading by more than an order of magnitude, from  $\sim 6 \,\mu$ m to  $\sim 300$  nm, mani-462 festing a significant grain refinement occurring within the ASBs. The 463 formation of ASBs is ascribed to the strain localization during the 464 465 impact loading, a phenomenon that has been commonly observed in 466 ultra-fine grained iron and copper after high-strain-rate deformation [34,38,39]. Although the formation of ASB is a localized plastic defor-467 mation process, which may help to dissipate the impact energy, the 468 grain refinement within the ASBs is clearly detrimental to fracture 469 resistance. Coarse-grain structures in nickel are known to induce 470 marked crack-tip blunting which can generate high toughness [23]; 471 however, once they are transformed into dynamically recrystallized 472 ultra-fine grains, the resistance to crack propagation becomes far lower. 473

Therefore, in contrast to the crack propagation in the NG $\rightarrow$ CG gradient 474 structure under quasi-static loading, where the growing crack was 475 arrested as it entered the coarser-grained region of the gradient [23], 476 under impact loading the brittle-like cracking mode that governed the 477 fracture process within the nano-grain zone continues to dominate throughout the originally coarser-grained end of the NG $\rightarrow$ CG gradient 479 zone due to this recrystallized grain refinement process. 480

The brittle-like fracture mode revealed here for the NG $\rightarrow$ CG gra-481 dient structure is also reflected in the load-displacement response of 482 the NG $\rightarrow$ CG specimen (Fig. 5) which shows limited energy adsorp-483 tion and a similar shape to that of the pure NG specimen. Moreover, 484 as the NG $\rightarrow$ CG structure has a lower strength than the NG structure 485 (Fig. 4 and Table 1), the maximum load achieved during the impact 486 test was lower in the NG $\rightarrow$ CG specimens. Consequently, the NG $\rightarrow$ CG 487 gradient structure displayed lower absorbed energies for both the 488 crack initiation and crack propagation stages (Fig. 6b), leading to an 489 overall energetic toughness under dynamic loading that was even 490 lower than that of the pure NG structure (Fig. 6a). 491

### 3.5.4. Gradient CG $\rightarrow$ NG structures

Similar to the NG $\rightarrow$ CG specimen, impact-tested CG $\rightarrow$ NG specimens were bent significantly, resulting in a macroscopically Vshaped crack with a wide opening angle of  $\sim$ 90° (see the inset in the top right corner of Fig. 10). However, the CG $\rightarrow$ NG structure was far tougher. Unlike the overall straight crack path generated by the brittle-like fracture in the NG $\rightarrow$ CG specimen (Fig. 9), the crack-path profile at the (plane-strain) mid-section for the CG $\rightarrow$ NG specimen 499



**Fig. 10.** Crack-path profile captured on the mid-plane section in a  $CG \rightarrow NG$  gradient specimen. Ductile fracture by micro-voids coalescence that dominated the initial pure CG zone was sustained in the adiabatic shear bands (ASBs) that were formed in the latter NG zone of the  $CG \rightarrow NG$  gradient. In these ASBs, plasticity was enhanced due to grain coarsening induced by dynamic recrystallization under dynamic loading. Regions C1-C5 reveal the microstructure resulting from dynamic plastic deformation in the plastic-wake regions along the crack path up to the crack tip. Region C6 shows the coarsened lamellar structures developed in the ASBs formed within the original nano-grain zone near the ligament end. By comparison, the representative original grain structures corresponding to the deformed-microstructures in regions C1-C6 are captured in remote regions R1-R6 that are far away from the crack path. The dynamic crack-path profile is delineated in white dotted lines. The propagation direction of the dynamic crack path is denoted by the red dotted arrows. The ASB region formed in the uncracked ligament is circumscribed by blue dotted curves. The striking end of the Charpy impact specimen is delineated by green dotted line.

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## **ARTICLE IN PRESS**

Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

revealed typical geometrical features resulting from ductile fracture 500 (Fig. 10). Specifically, following crack initiation from the notch tip 501 within the pure CG zone, the extending crack tip became blunted due 502 to significant plastic deformation within the coarse-grained ( $\sim 8 \,\mu m$ ) 503 microstructure. As shown in Fig. 10, the blunting continued until the 504 crack formed semi-circular blunted crack tip at a region where the 505 original grain size was  $\sim$ 2.5  $\mu$ m (region C3 in Fig. 10). Close examina-506 tion of the microstructures in regions C1-C3 near the blunted crack 507 508 path revealed copious formation of dislocation slip bands, dislocation 509 substructures and cells, which again caused substantial grain refine-510 ment in the plastic-wake regions. The depth of the impact-affected plastic zone surrounding the blunted crack profile was estimated to 511 be as large as  $\sim$ 300  $\mu$ m, which significantly contributes to the high 512 levels of absorbed energy during the stages of crack initiation and 513 514 stable crack growth in this gradient structure (Fig. 6b and Table 2).

When the crack propagated further into the gradient  $CG \rightarrow NG$  zone, 515 microvoids nucleated and coalesced, leading to the initial development 516 of a mode-I crack path, as shown in region C2 as well as the main crack 517 518 path subsequent to the blunted profile in Fig. 10. However, once the 519 crack proceeded into regions of the gradient where the grain size decreased to below  $\sim 1 \,\mu$ m, the prior mode-I crack path was no longer 520 sustained and instead followed a distinctly curved trajectory. Repeat-521 edly, microvoids were formed in the plastic-wake regions along the 522 crack path indicative of crack extension by microvoid coalescence asso-523 524 ciated with extensive plastic deformation (e.g., in region C4 in Fig. 10). The curved cracking eventually terminated in the nano-grained zone of 525 the gradient where the original grain size was  $\sim$ 80 nm, leaving a 400-526  $\mu$ m uncracked ligament. Immediately ahead of the arrested crack tip, in 527 region C5 in Fig. 10, a lamella grain structure elongated in the direction 528 of the crack path was developed. Compared to the  $\sim$ 80 nm width of the 529 original columnar grains at this location, the width of the elongated 530 531 grains varied from 100 to 800 nm, clearly showing dynamic grain coarsening had occurred in this region. Within this narrow deformation 532 533 region, which extended from the crack tip to the bottom (impacted) surface of the specimen (bounded by blue dotted curves in Fig. 10), there 534 was evidence of abundant adiabatic shear bands. However, unlike the 535 dislocation substructures that formed in the initial pure CG zone, this 536 latter region of the CG $\rightarrow$ NG gradient structure comprised fine lamella-537 538 shaped grains, several micrometers in length sized with a thickness that 539 varied between  $\sim$ 50 to 300 nm (see region C6 in Fig. 10).

The coarsening of the nano-grains and associated formation of 540 ASBs near the end of the CG $\rightarrow$ NG gradient zone can again be attrib-541 uted to dynamic recrystallization during the impact loading. Such 542 543 grain coarsening in the ASBs leads to structural softening which can promote plasticity in the otherwise high-strength nano-grained 544 region [11,40]. Thus, unlike the pure NG structure where cracking 545 proceeded by a brittle-like mode, the enhanced plasticity associated 546 with the coarsening of the nano-grain zone in the  $CG \rightarrow NG$  gradient 547 acts to preserve a ductile mode of crack advance. This led to the load-548 displacement response of the  $CG \rightarrow NG$  specimen showing a pro-549 longed softening tail, with a total displacement similar to that of the 550 551 pure CG specimen (Fig. 5). Moreover, as the tensile strength of the 552  $CG \rightarrow NG$  structure is ~30% higher than that of the pure CG structure (Fig. 4), the maximum load achieved during impact loading was also 553 correspondingly higher by  $\sim$ 100%. The result is that CG-NG gradient 554 structure displays by far best crack resistance to impact loading over 555 all the other monolithic and gradient structures, with a higher impact 556 557 energy coupled with a tensile strength exceeding 750 MPa.

#### 558 4. Discussion

#### 559 4.1. Estimation of the dynamic J-R curve

To further evaluate the fracture resistances of the gradient- and monolithic- structured Ni under the impact loading, we estimated the nonlinear-elastic fracture-mechanics-based dynamic *J*-R curves



**Fig. 11.** Dynamic *J*-R curves, showing the variation in the value of *J* to sustain crack extension  $\Delta a$ , for the studied monolithic and gradient Ni structures under impact loading. The R-curves are estimated from the load-displacement data (Fig. 5) using the key curve method [41,42]. Interestingly, dependent on the direction of crack growth with respect to the gradient in grain size, the gradient structures display the best (for the CG $\rightarrow$ NG gradient) of the Ni structures evaluated.

from the load-displacement data following the key curve method 563 [41,42]; the resulting crack-resistance curves are shown in Fig. 11. 564 Details of the key curve method are described in Appendix A. 565

During the initial stages of crack growth ( $\Delta a < \sim 0.2 \text{ mm}$ ), the 566  $NG \rightarrow CG$  gradient structure, akin to the two monolithic NG and CG 567 structures, required a low applied J-value to instigate and sustain crack-568 ing. In contrast, the corresponding applied J to sustain initial cracking 569  $(\Delta a < \sim 0.2 \text{ mm})$  in the CG $\rightarrow$ NG gradient specimen was roughly two 570 times higher than those in the uniform coarse-grain CG and  $NG \rightarrow CG$ 571 specimens. As the crack further extended (in excess of  $\Delta a \sim 0.2$  mm), 572 the dynamic *I*-R-curves of the  $CG \rightarrow NG$  and CG specimens rose very 573 steeply. The R-curves for the  $CG \rightarrow NG$  gradient structure maintained the 574 highest crack resistance of all the Ni structures tested. By comparison, 575 the uniform-grained NG and the NG $\rightarrow$ CG gradient specimens, displayed 576 only marginally rising R-curves with crack extension, a characteristic 577 that is indicative of low resistance to crack propagation and hence infe-578 rior damage-tolerance. 579

The rising J-R curves for all the structures terminated as the crack 580 extended to  $\sim 0.6$  mm, after which the R-curves (except for the lowest 581 NG $\rightarrow$ CG R-curve) started to drop. The maximum J-integral value achieved in the CG $\rightarrow$ NG gradient specimen was  $\sim 800$  kJ/m<sup>2</sup>, which is 583 roughly 1.6 times as that in ductile CG specimen and 3 times higher 584 than those in the NG and the NG $\rightarrow$ CG gradient specimen. 585

The dynamic *I*-R curves results presented in Fig. 11 confirm that the 586 impact toughness has a strong dependency on the gradient orientation 587 in GS Ni. Negligible toughening is observed in the NG $\rightarrow$ CG gradient 588 structure as revealed by the insignificant-rising dynamic J-R curve, 589 implying a typical brittle fracture behavior akin to the pure NG structure. 590 However, the  $CG \rightarrow NG$  gradient structure displays an exceptional rising 591 *I*-R curve, suggesting a dynamic fracture toughness superior to all the 592 investigated GS and MS structures. 593

### 4.2. Extraordinary strength-impact toughness synergy in gradientgrained structures

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For conventional structural materials, there is invariably an 596 unavoidable trade-off between strength and fracture toughness, 597 which is clearly demonstrated by the Ashby map where the yield 598 strength vs. the fracture toughness are plotted [43]. As in nature, one 599 approach to solving this strength-toughness "conflict" is through the 600 use of gradient-grained architectures [9]. In line with this, we have 601 successfully synthesized bulk GS-Ni structures possessing a gradient 602 in grain size in two configurations (*i.e.*, NG $\rightarrow$ CG gradient and CG $\rightarrow$ NG 603 gradient). Our earlier work has revealed that the best combination of 604

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**Fig. 12.** Strength *vs.* impact toughness. (a) Plot showing the total absorbed impact energy as a function of the quasi-static tensile strength. (b) Ashby map showing the approximate dynamic crack-growth toughness  $K_{ss}$  at  $\Delta a \sim 0.5$  mm as a function of the quasi-static tensile strength for the GS and MS structures; here,  $K_{ss} = (E' f)^{1}/_{2}$ ,  $E' = E/(1-v^{2})$ . For pure Ni, *E* and v are 200 GPa and 0.312, respectively.

strength and toughness properties can be achieved under the static loading with a coarse- to fine-grained (CG $\rightarrow$ NG) gradient structures [23]. Here, we extend this approach to examine whether the strength-toughness synergy still prevails with such gradient structures under impact/dynamic loading conditions.

The impact toughness of the GS and MS structures are plotted in 610 Fig. 12 as a function of their tensile strengths. Two measurements of 611 612 the impact toughness are utilized, i.e., a global measure using the total absorbed impact energy (W) (Fig. 12a), and a nonlinear-elastic 613 614 fracture-mechanics-based estimate of the dynamic crack-growth toughness ( $K_{ss}$ ) at  $\Delta a \sim 0.5$  mm (Fig. 12b). (Note that the method 615 used to convert the estimated J-based crack-growth toughness values 616 into approximate K-based values is described in the Appendix A). 617 As presented in Fig. 12a, there exists a trade-off between the 618 619 strength and the total absorbed impact energy for the CG, NG, 620 and NG $\rightarrow$ CG gradient structures, as shown by the light gray zone in Fig. 12a. Specifically, although the CG specimens have the low-621 est tensile strength, the total absorbed impact energy of the CG 622 specimens is almost  ${\sim}80\%$  higher than those of the NG ${\rightarrow}$ CG gra-623 dient and the NG specimens. However, this issue of the generally 624 mutually exclusive properties of strength and toughness can be 625 626 alleviated with a coarse- to nano-grained (CG $\rightarrow$ NG) gradient structure which, compared to the uniform CG material, displays 627 simultaneous increases in tensile strength and total absorbed 628 impact energy by, respectively, 125% and 60% (as shown by the 629 red data point above the midst of the light gray zone in Fig. 12a). 630 631 Indeed, the CG $\rightarrow$ NG structure is clearly superior to the pure NG 632 and even NG $\rightarrow$ CG gradient structures, as shown in Fig. 12b. The 633 fascinating consequence of this is that although gradient grain-sized structures can combine the high strength of the nano-grained struc-634 tures and the far higher toughness of the coarser-grained ones, this 635 is only the case under impact loading if the crack grows in the gen-636 eral direction of a coarse-to-fine grained gradient, i.e., for the pres-637 638 ent Ni CG $\rightarrow$ NG structure, due to the beneficial effects of nano-grain coarsening by dynamic recrystallization. 639

640 4.3. Gradient-orientation-dependent impact toughness: implication for641 engineering applications

The fracture resistance to the impact loading in the gradient structured materials is strongly dependent on the gradient orientation. As clearly observed in Fig. 12b, the crack-growth toughness ( $K_{ss}$ ) (398 MPa·m<sup>1/2</sup>) at  $\Delta a \sim 0.5$  mm of the CG $\rightarrow$ NG structure far surpasses that (183 MPa·m<sup>1/2</sup>) 645 in the NG $\rightarrow$ CG structure.<sup>2</sup> This finding is consistent with the difference of 646 the absorbed impact energies during the various stages of fracture. For 647 ductile materials subjected to impact loading, the absorbed energy for 648 crack growth ( $W_{\text{stable}} + W_{\text{unstable}}$ ), rather than that for crack initiation 649  $(W_{in})$ , contributes to most of the total absorbed energy. Accordingly, the 650 impact toughness is more likely to be associated with the resistance to 651 crack growth rather than the resistance to crack initiation during the 652 impact loading. As revealed in our study, the values of  $W_{\text{stable}}$  and  $W_{\text{unstable}}$ 653 in the  $NG \rightarrow CG$  specimens are considerably lower than those in the 654  $CG \rightarrow NG$  specimens (Fig. 6b), representative of the fact that the impact 655 resistance of the CG $\rightarrow$ NG gradient structure is far superior to that of the 656  $NG \rightarrow CG$  gradient structure. 657

With respect to the material structure design for engineering 658 application, our current findings suggest that create fine- to coarse-659 grained gradient architectures may not be optimal for a material's 660 impact toughness, although we have not found such behavior under 661 quasi-static loading conditions [23]. Actually, under the quasi-static 662 loading, the slope of the quasi-static J-R curve keeps rising as the 663 crack grows from the NG region to the CG region, indicating an 664 enhanced crack-growth toughness. Therefore, the NG→CG gradient 665 structure might be preferred for certain safety-critical applications 666 under the quasi-statistic loading conditions, but should be avoided in 667 applications that involve high strain-rate loading. 668

On the other hand, as shown in Fig. 12b, the crack-growth tough-669 ness of the CG $\rightarrow$ NG specimens are much higher than the NG $\rightarrow$ CG 670 specimens and even surpass that of the CG specimens. This indicates 671 that the CG $\rightarrow$ NG gradient structure is more suited for dynamic load-672 ing conditions. As confirmed from the analysis of the absorbed energies (Fig. 6), the CG $\rightarrow$ NG gradient structure significantly increases 674 the absorbed energies for both crack initiation and unstable growth, 675 without losing resistance to the stable crack growth. In short, coarseto fine-grained gradient structures present distinct advantages over 677

<sup>&</sup>lt;sup>2</sup> For  $K_{SS}$  (or  $J_{SS}$ ) to be considered as a size-independent fracture toughness value, the validity requirements for the *J*-field dominance and plane-strain conditions should be

met [22], *i.e.*, that  $b_0$ ,  $B > \frac{10f_{ss}}{\sigma_0} = \frac{10K_{ss}^2/(E/(1-v^2))}{\sigma_0}$ , where  $\sigma_0$  is the flow strength (average of yield strength and tensile strength).  $b_0$  and B are the initial ligament length and specimen thickness, respectively. The  $K_{ss}$  values measured for the NG, NG→CG, NG→CG, and CG specimens (Fig. 12b) are 222 MPa·m<sup>1/2</sup>, 183 MPa·m<sup>1/2</sup>, 398 MPa·m<sup>1/2</sup>, and 315 MPa·m<sup>1/2</sup>, respectively. As all of these values do not satisfy the ASTM requirements for *J*-dominance at the crack tip, these values are not strictly ASTM valid [22].

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Y. Lin et al. / Acta Materialia xxx (2020) xxx-xxx

uniform grained structures in terms of their superior damage-toler-678 679 ance as structural materials for applications involving high-strainrate impact loading conditions. 680 681

#### 682 5. Conclusions

683 Based on an experimental study of the effects of grain-size gradients on the strength and dynamic fracture resistance of bulk nickel 684 plates, the following conclusions can be made: 685

686 1. Using gradients in grain size, synthesized by direct current electrodeposition, from nano-sized grains of ~30 nm (NG) to coarser 687 688 grains of ~8  $\mu$ m (CG), the impact toughness of the gradient-689 structured (GS) Ni, assessed using instrumented Charpy impact 690 testing, was compared to that of uniform grain-sized nano-691 grained (NG) and coarse-grained (CG) structures. Gradient orientation was found to significantly affect the impact toughness. 692 Test specimens with an NG $\rightarrow$ CG gradient structure displayed a 693 relatively low fracture resistance to impact loading, similar to 694 that exhibited by the nominally brittle uniform nano-grained 695 (NG) specimens. Conversely, although the tensile strength of the 696 gradient structures was 43% lower than that of the NG structures, 697 the  $CG \rightarrow NG$  gradient structure displayed a strength and impact 698 699 energy that was over  $\sim$ 2.0 and  $\sim$ 1.5 times higher than those of the CG structure. Indeed, the CG $\rightarrow$ NG gradient structure was 700 definitively the most damage-tolerant Ni structure representing 701 the best combination of strength and impact toughness. 702

703 2. The superior dynamic fracture resistance of the CG $\rightarrow$ NG gradient structure was ascribed to the sustained ductile fracture by micro-704 void coalescence, which was active not only in the initial coarse-705 grained region, but also in the adiabatic shear bands formed in 706 707 the subsequent nano-grained region, as plasticity was enhanced in the latter region due to the grain coarsening induced by 708 709 dynamic recrystallization under the impact loads.

710 3. This work presents a novel strategy of using coarse- to nano-711 grained gradient structures in order to design strong, tough and 712 hence damage-tolerant metallic structures for engineering appli-713 cations involving high-strain rate/impact loading.

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#### Supplementary materials 721

Supplementary material associated with this article can be found 722 723 in the online version at doi:10.1016/j.actamat.2020.04.027.

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#### 725 Appendix A. Estimation of dynamic J-R curves

The nonlinear-elastic fracture-mechanics-based J-integral resis-726 tance curve (i.e., the J-R curve) is commonly used to describe the frac-727 ture resistance as a function of crack extension [22,37]. Due to its 728 729 effectiveness in characterizing both the crack-initiation and crackgrowth toughnesses of ductile materials, the *I*-integral and *I*-R curve 730 have become important material parameters in the evaluation of the 731 732 damage-tolerance of structural materials. As these parameters can be similarly utilized at high loading rates, the pre-cracked impact speci-733 mens can be utilized to determine the dynamic J-R curve from the 734 recorded load-displacement curves using the instrumented Charpy 735 impact testing [26]. For the notched impact specimens without 736 fatigue pre-cracking, which were used in our current study, prior 737

work has shown that the dynamic J-R curve can still be estimated 738 using the so-called "key curve" method [41,42].<sup>3</sup> 739

Based on this method, the load-displacement curve between  $P_{gy}$ 740 (load at yield) and  $P_{in}$  (load at crack initiation) can be fitted by the fol-741 lowing power law: 742

$$\frac{PW}{b_0^2} = k \left(\frac{\Delta_{\rm pl}}{W}\right)^m,\tag{A1}$$

where *P* and *W* denote the load and the specimen width ( $\sim$ 3 mm). 744 respectively;  $a_0$  (~1.5 mm) and  $b_0 = (W - a_0)$  are the initial crack 745 length (equal to the notch depth in this work) and the initial ligament 746 length, respectively.  $\Delta_{pl} = \Delta - C_{el} P$  refers to the displacement due to 747 plastic deformation, where  $\Delta$  is the instant displacement while  $C_{el}$  is 748 the elastic compliance, determined as the slope  $(P_{\rm el}/\Delta_{\rm el})$  of the linear 749 elastic section of the load-displacement curve. Once the dimension-750 less constants k and m are determined, Eq. (A1) can be applied to 751 derive the remaining ligament length (*b*) from the load-displacement 752 curve beyond the crack initiation point using following relationship 753 [41]: 754

$$\frac{PW}{b^2} = k \left(\frac{\Delta_{\rm pl}}{W}\right)^m,\tag{A2}$$

Accordingly, the crack extension  $\Delta a$  at each point after crack initi-756 ation can be derived: 757

$$\Delta a = b_0 - b = W - a_0 - b = (W - a_0) - \sqrt{\frac{PW^{m+1}}{k(\Delta_{pl})^m}}$$
(A3)  
758

The dynamic *I*-integral value can then be calculated from the rela-759 tionship for pure bending proposed by Rice and coworkers [44]: 760

$$J = \frac{\eta U}{B(W - a_0)},\tag{A4}$$

where *B* is the specimen thickness ( $\sim$ 5 mm), *U* is the potential energy 762 during the impact testing, which equals to the total absorbed energy 763 determined by finding the total area under the load-displacement 764 curve. The value of *n* was taken as 2 for pre-cracked and 1.46 for 765 Charpy V-notch specimen [45]. Considering the effect of crack growth 766 on the *I*-integral, the Garwood's formula [46] to correct the dynamic 767 J-integral values for the three-point bend specimen was used: 768

$$J_n = J_{n-1} \frac{W - a_n}{W - a_{n-1}} + \frac{\eta U_n}{B(W - a_{n-1})},$$
(A5)
  
769

where  $U_n$  refers to the area under the actual test record between deflec-770 tion  $\Delta_n$  and  $\Delta_{n-1}$ . By following Eqs. (A1) to (A5), we can construct the 771 dynamic *J*-R( $\Delta a$ ) curve by calculating the *J*-integral corresponding to the 772 specific crack extensions ( $\Delta a$ ) on the basis of the recorded load-displace-773 ment/deflection curve. The off-set power law was used to fit the dynamic 774 *J*-R curves at the stage of stable crack growth [45]: 775

$$J = m + l(\Delta a)^n, \tag{A6}$$

where *m*, *l*, and *n* are fitting parameters.

777 Approximate stress intensity K-based toughness values were 778 back-calculated from the J values using the standard mode-I J-K 779 equivalence relationship: 780

$$J = K^2 / E',$$
 (A7) <sub>781</sub>

<sup>&</sup>lt;sup>3</sup> The J-integral, like other characterizing parameters in fracture mechanics such as the stress intensity factor K, were originally derived for homogeneous isotropic continua [47]. However, they have been widely used, indeed mostly used, to examine materials and microstructures where such homogeneity is difficult to rationalize. The underlying justification for using such fracture mechanics analyses is that the structural size-scales remain small compared to the extent of the J-dominated (or K-dominated) crack-tip stress and displacement fields, and that this in turn remains small compared to the macroscale of the test sample

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where is  $E' = E / (1 - v^2)$  for plane-strain conditions; *E* is Young modulus and v is Poisson's ratio. 782

- 783 For the purposes of comparison of the effective fracture tough-
- nesses of the various monolithc and gradient Ni structures in Fig. 12, 784
- a crack-growth toughness for dynamic loading, K<sub>ss</sub>, was calculated as 785
- the approximate K values for a crack-extension of  $\Delta a = 0.5$  mm. 786

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